



NUCLEAR PROPULSION – Conclusions from the Workshop on Nuclear Fission for Future Energy and Space Application: Thin Film and Other Unconventional Solution”

On the 10th and 11th May 2002, at the University of Rome “La Sapienza”, a Workshop on Nuclear Propulsion was held. This Workshop was sponsored by ESF-European Science Foundation, with the title: “ESF-PESC Exploratory Workshop on Nuclear Fission for Future Energy and Space Application: Thin Film and Other Unconventional Solution”.

The purpose was to review and discuss different propulsion systems based on advanced or innovative nuclear technology. Such propulsion is indispensable for interplanetary missions, as it has the potential for reducing overall mission cost and increased flexibility. The Workshop was attended by about 30 people from EU, USA and Israel, representing industries (Fiat Avio – Italy, AEA Technology – UK, Technicatome – France, etc.) and institutions (Carlo Rubbia – ENEA, ESA, NASA, St. Louis University, Ben Gurion University etc.). Collaboration between industry and academia was recognized key to future enterprise in this area. International collaboration was mentioned as essential and definitely wished, seeing the complexity, cost, safety considerations and standardisation of any nuclear reactor used for space propulsion. Much of the discussion during the second day of the Workshop was devoted to ways and means of jumpstarting initiatives in this field.

During the Workshop a spectrum of proposals and ideas were presented; they can be grouped in three main areas.

1) Direct conversion of fission heat into thrust

The thin fissioning film solution (“Rubbia’s engine”) was presented and discussed by Carlo Rubbia (Nobel Prize in physics, 1984) himself. The role of Am²⁴² was pointed out, and above all the fact that only Am²⁴² may achieve criticality in a thin film state (0.2 microns is the thickness that was calculated, so that the fission product cannot be trapped in the fuel). Prof. Y. Ronen-Ben Gurion University described a synthetic history of thin layer fission studies in Israel, leading to conclusions similar to those of C. Rubbia as to the feasibility of using Am fission as an energy source for interplanetary travel or even terrestrial power generation. He showed results of calculations indicating ways Am²⁴² may be obtained using a fast reactor: his results are very encouraging, and pave the way to substantial production of Am²⁴², i.e. far beyond what is needed to perform experiments. In the context of doing the experiments needed to validate his engine, C. Rubbia proposed the use of a cyclotron to simulate the behaviour of such a scheme in a test facility. Ar ions created by spallation of protons would be the ersatz fission fragments family simulating actual fission products.

Nuclear Propulsion Projects developed in the past were presented by D. Froning – Flight Unlimited. In these systems the temperature and performance (differently from Rubbia’s engine) is limited by the fact that the heat is transmitted through the reactor solid core (e.g., NERVA in US). Only Pebble Bed Reactors seemed to give higher performance due to the very nature of the topology of heating.



2) Nuclear Power used to produce directly electricity (so called NEP)

The concept system presented by Robert Bond (AEA Technology) exploits the heat from nuclear reactions to produce electricity in a thermodynamic cycle. The electricity is used in electromagnetic acceleration systems such as: MHD, RF heated plasma, ion propulsion, to produce thrust.

A. Ilin and A. Petro (NASA-JSC) presented the VASIMR MPD propulsion system. In the first pilot of this engine RF-heated plasma is expected to give a specific impulse $\approx 10\,000$ s with a thrust of 0,2 N. Questions on the plasma acceleration and radiative losses were raised by the audience.

Paul Czysz (St. Louis University) presented past experiments conducted at WPAFB to test high temperature gas and plasma acceleration by means of trains of RF pulses to heat the plasma. The main purpose at the time was to try replace conventional wind tunnels with MHD-accelerated wind tunnels.

3) Hybrid Systems

Methods that allow to increase the specific impulse obtained by nuclear thermal were discussed too. The use of low frequency (10 KHz) of RF inductive heating of H₂ or other gas propellant was presented by C. Dujarric-ESA. Injection of liquid oxygen (“post combustion”) could be possible in his concept, to increase thrust momentarily, as when attempting takeoff from the surface of a planet or satellite. Furthermore it was argued that the thermodynamic efficiency of the Power Generator would be increased by the temperature ratio between liquid hydrogen (before entering the nuclear heater) and the reactor temperature.

An important part at the Workshop were the free discussions had by participants both the first day, during scheduled talks, and the second day, explicitly dedicated to ‘General Discussion and Roadmap Proposal’.

From these discussions some consensus emerged. Though nuclear technology for space mission presents some minuses, e.g., public acceptance, safety, application range, nevertheless it has some unique features that must be considered. The use of nuclear propulsion allows wider mission ranges, enables shorter time, yields high power per given mass, in one word it gives much more **flexibility**. Dual use (e.g. power generation) gives a further reason to develop this technology.

J-P. Roux-Technicatome basing on experiences acquired in nuclear generators for submarines, proposed a roadmap, with a tentative kick off in 2003. The proposed work would start testing in orbit conventional NP, while studying 1 or 2 innovative and advanced NP concepts for a 2025 manned Mars mission.



FINAL PROGRAMME OF THE WORKSHOP

10th MAY

- 09.00-09.10 ESF introduction – Prof. Laurent
- 09.10-09.20 Welcome – Prof. Bruno
- 09.20-10.30 Rubbia
- 10.30-11.00 Ronen - Ben Gurion University
- 11.00-11.15 Discussion
- 11.15-11.30 Coffee Break
- 11.30-12.00 Bond - AEA Technology Space
- 12.00-12.30 Froning - Flight Unlimited
- 12.30-12.45 Discussion
- 12.45-13.45 Lunch Break
- 13.45-14.15 Dornbusch/Roux - Technicatome
- 14.15-14.45 Dujarric - ESA
- 14.45-15.00 Discussion
- 15.00-15.30 Czysz - St. Louis University
- 15.30-16.15 Ilin/Petro - NASA Johnson Space Center
- 16.15-17.15 Comments from ESA
- 17.15-18.15 Comments from Industries

11th MAY

- 9.00-12.00 General Discussion and Roadmap Proposals



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Total number of participants: 24

13 from Italy, 5 from USA, 3 from France, 1 from UK, 1 from Germany, 1 from Israel

Age range: 29 – 70

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